PAGE: 1 PRINT DATE: 05/08/96

FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL HARDWARE

NUMBER: M5-6MB-2078-G -X

SUBSYSTEM NAME: ELECTRICAL POWER GENERATION - CRYO, GENERIC

REVISION: 9 09/09/92

PART DATA

**PART NAME** 

**VENDOR NAME** 

PART NUMBER

VENDOR NUMBER

LRU

: PANEL R1A2

V070-730276

SRU : RESISTOR

RWR80\$1211FR

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

RESISTOR, CURRENT LIMIT, 1.2K OHM, 2 WATT - H2 MANIFOLD VALVE, TANKS 1 AND 2

REFERENCE DESIGNATORS:

32V73A1A2A1R3

32V73A1A2A1R7 32V73A1A2A1R12 32V73A1A2A1R15

QUANTITY OF LIKE ITEMS: 4

FOUR, TWO PER H2 MANIFOLD VALVE CIRCUIT

FUNCTION:

PROVIDES CURRENT LIMIT/CIRCUIT PROTECTION FOR THE CONTROL CIRCUIT OF H2 MANIFOLD ISOLATION VALVES 40V45LV031 AND 40V45LV041.

### FAILURE MODES EFFECTS ANALYSIS FMEA -- CIL FAILURE MODE

NUMBER: M5-6MB-2078-G- 01

REVISION#: 9

04/16/96

SUBSYSTEM NAME: ELECTRICAL POWER GENERATION - CRYO, GENERIC

LRU: PANEL R1A2 ITEM NAME: RESISTOR **CRITICALITY OF THIS** FAILURE MODE: 1R2

FAILURE MODE:

OPEN

MISSION PHASE:

LO LIFT-OFF

00 ON-ORBIT DO DE-ORBIT

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY

104 ATLANTIS

105 ENDEAVOUR

CAUSE:

STRUCTURAL FAILURE (MECHANICAL STRESS, VIBRATION), ELECTRICAL STRESS, THERMAL STRESS, PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) FAIL

C) PASS

PASS/FAIL RATIONALE:

A)

REDUNDANCY SCREEN "B" FAILS EVEN THOUGH THE FAILURE OF THIS RESISTOR IS DETECTABLE BECAUSE THE TIME FOR CORRECTIVE ACTION (ELECTRICAL LOAD) RECONFIGURATION) EXCEEDS THE TIME TO EFFECT (MANIFOLD GROSS EXTERNAL LEAK STARVES TWO FCP'S DURING ASCENT/DESCENT).

C)

- FAILURE EFFECTS -

(A) SUBSYSTEM:

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FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL FAILURE MODE

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LOSS OF ABILITY TO OPEN AFFECTED MANIFOLD VALVE AFTER INADVERTENT OR COMMANDED VALVE CLOSURE.

(B) INTERFACING SUBSYSTEM(S): SAME AS (A)

#### (C) MISSION:

(CRIT 2/2) POSSIBLE LOSS OF MISSION DUE TO ASSOCIATED MANIFOLD VALVE FAILING CLOSED RESULTING IN ONE TANK BEING ISOLATED TO A SINGLE FUEL CELL. MISSION TERMINATED WHEN THE HYDROGEN IN THAT TANK IS CONSUMED.

(D) CREW, VEHICLE, AND ELEMENT(S): NO EFFECT - FIRST FAILURE

## (E) FUNCTIONAL CRITICALITY EFFECTS:

(CRIT 1R2) POSSIBLE LOSS OF CREW/VEHICLE DUE TO THE FOLLOWING SCENARIO: 1) RESISTOR OPENS (VALVE REMAINS OPEN), AND 2) GROSS EXTERNAL LEAK STARVES TWO FCP'S (LOSS OF TWO FCP'S DURING ASCENT LOSES CREW/VEHICLE, LOSS OF A SECOND FCP DURING DESCENT LOSES CREW/VEHICLE IF INSUFFICIENT TIME IS AVAILABLE FOR AN ELECTRICAL LOAD RECONFIGURATION RESULTING IN THE INABILITY OF THE SINGLE REMAINING FUEL CELL TO SUPPLY ADEQUATE ELECTRICAL POWER.)

#### -DISPOSITION RATIONALE-

#### (A) DESIGN:

REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR

### (B) TEST:

GROUND TURNAROUND TEST

ANY TURNAROUND CHECKOUT TESTING IS ACCOMPLISHED IN ACCORDANCE WITH OMRSD. THE OMRSD DATA PROVIDED BELOW IS NO LONGER BEING KEPT UP-TO-DATE. IF THERE IS ANY DISCREPANCY BETWEEN THE GROUND TESTING DATA PROVIDED BELOW AND THE OMRSD, THE OMRSD IS THE MORE ACCURATE SOURCE OF THE DATA.

RESISTOR INTEGRITY IS FUNCTIONALLY VERIFIED DURING FLIGHT. PERFORM GROUND TURNAROUND TEST WHEN VALID VERIFICATION IS UNOBTAINABLE IN FLIGHT OR AFTER LRU REPLACEMENT.

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# FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL FAILURE MODE

NUMBER: M5-6MB-2078-G-01

(C) INSPECTION:

REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR

#### (D) FAILURE HISTORY:

CURRENT DATA ON TEST FAILURES, FLIGHT FAILURES, UNEXPLAINED ANOMALIES, AND OTHER FAILURES EXPERIENCED DURING GROUND PROCESSING ACTIVITY CAN BE FOUND IN THE PRACA DATA BASE. THE FAILURE HISTORY DATA PROVIDED IN APPENDIX & IS NO LONGER BEING KEPT UP-TO-DATE.

#### (E) OPERATIONAL USE:

NO CREW ACTION AFTER FIRST FAILURE.

### - APPROVALS -

PAE MANAGER

PRODUCT ASSURANCE ENGR

DESIGN ENGINEERING EDITORIALLY APPROVED

TECHNICAL APPROVAL

: P. STENGER-NGUYEN :/

; J. NGUYEN

: T. D. NGUYEN

: JSC

: VIA APPROVAL FORM : 96-CIL-012\_M5-6MB